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GENERAL MOTORS CORP INDIANAPOLIS IN DETROIT DIESEL A--ETC F/G 21/5
MARITIME PATROL AIRCRAFT (MPA) CONCEPT FORMULATION. ALLISON PD3--ETC(U)
FEB 79 P STOLP N62269-78-C-0415
DDA-EDR-9774A

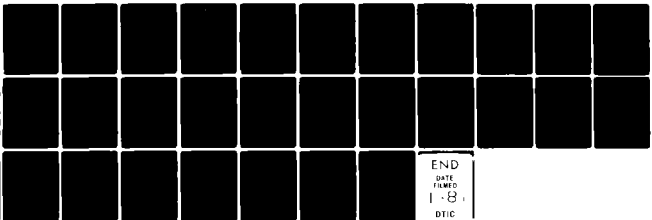
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19 REPORT DOCUMENTATION PAGE		READ INSTRUCTIONS BEFORE COMPLETING FORM
1. REPORT NUMBER 18 NADC-80181-B-60	2. GOVT ACCESSION NO. AD-A093	3. RECIPIENT'S CATALOG NUMBER 276
4. TITLE (and Subtitle) Maritime Patrol Aircraft (MPA) Concept Formulation. Allison PL37A-37. T702 Derivative Turboprop Engine.		5. TYPE OF REPORT & PERIOD COVERED 9 Final Report Oct 1978 - Feb 1979
7. AUTHOR(s) 12 P./Stolp		6. PERFORMING ORG. REPORT NUMBER (14) DDA-EDR-9774A
9. PERFORMING ORGANIZATION NAME AND ADDRESS Detroit Diesel Allison Division of General Motors Corporation, Box 894 Indianapolis, IN 46206		8. CONTRACT OR GRANT NUMBER(s) (15) N62269-78-C-0415
11. CONTROLLING OFFICE NAME AND ADDRESS Naval Air Development Center Warminster, PA 18974		10. PROGRAM ELEMENT, PROJECT, TASK AREA & WORK UNIT NUMBERS 65152N A03P-03PA/001E/7W0880-001 Work Unit XM201
14. MONITORING AGENCY NAME & ADDRESS (if different from Controlling Office) (16) W0220 (17) W0880002		12. REPORT DATE February 1979
		13. NUMBER OF PAGES 34 (12) 36
		15. SECURITY CLASS. (of this report) UNCLASSIFIED
		15a. DECLASSIFICATION/DOWNGRADING SCHEDULE
16. DISTRIBUTION STATEMENT (of this Report) APPROVED FOR PUBLIC RELEASE; DISTRIBUTION UNLIMITED		
17. DISTRIBUTION STATEMENT (of the abstract entered in Block 20, if different from Report)		
18. SUPPLEMENTARY NOTES		
19. KEY WORDS (Continue on reverse side if necessary and identify by block number) Aircraft Propulsion Turbofan Engine Derivative Engine Turboshaft Engine Maritime Patrol Aircraft Turboprop Engine		
20. ABSTRACT (Continue on reverse side if necessary and identify by block number) This study developed data on Detroit Diesel Allison (DDA) common core derivative engines for use in Maritime Patrol Aircraft (MPA) concept formulation studies. The study included the screening of potential DDA turboprop/turboshaft engines and the preparation of technical and planning information on three of the most promising engine candidates plus an all new engine. Screening of DDA derivative candidates was performed utilizing an analytical MPA model using synthesized mission profiles to rank the candidates in terms		

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of fuel consumption, weight, cost and complexity. The three turboprop engines selected for further study were as follows: a derivative of the unity size T701-AD-700 shaft power engine with rematched turbine (PD 370-37), an advanced T701 turboprop derivative with 25:1 overall pressure ratio and a scaled ATEC demonstrated compressor (PD 370-40), an advanced T701 turboprop derivative with 17.7:1 overall pressure ratio and a scaled ATEGG demonstrated compressor (PD 370-41). Data is also presented on a new advanced turboprop engine with 30:1 overall pressure ratio which incorporates compressor, combustor, turbine, and cooling technology now under development and demonstration at DDA. The documentation consists of six separate reports prepared in the following manner. One report summarizes the engine screening analysis and describes the approach to, and the conclusions of the study. A separate report for each of the three derivative engines and for the new turboprop present estimates of performance, weight, and dimensional data. The engineering budgetary estimates of the development, acquisition, and service costs for each of the four engines are presented in a separate report.

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REVISIONS

<u>Letter</u>	<u>Page</u>	<u>Revision</u>
A	4	Gearbox and total weight
A	5	Gearbox, interconnecting struts and shaft, and total weight
A	9	Additional matrix points at 0 and 25,000 feet
A	12	Additional performance
A	17	Additional performance
A	22	Additional performance
A	27	Additional performance



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I. INTRODUCTION

This report presents estimates of performance, weight, and dimensional data for the PD370-37 turboprop engine. The engine is in the 8000 to 9000 SHP class and is intended for subsonic aircraft. The data is submitted for use in preliminary design type studies in the evaluation of turboprop systems.

The PD370-37 is a derivative of the unity size T701-AD-700 shaft power engine, with the turbine rematched for better turboprop characteristics. The T701-AD-700 is a free turbine turboshaft engine that was developed through safety demonstration testing, for the U.S. Army's HLH program. The Model 570, a commercial industrial version of this engine, has undergone additional development testing, and is now in production.

The reduction gearbox for speed reduction to the prop-fan is a new simplified design compared to the DDA T56 series of gearboxes. The new design is based upon a study into the reliability and maintenance cost history of past turboprop systems, and follows the recommendations from that study for a gearbox with high reliability, easy maintainability, and low maintenance cost.



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II. ENGINE DESCRIPTION

The Model PD 370-37 is a single spool, free power turbine, axial flow power unit connected by shafting, and supporting structure to an offset reduction gear assembly. The general arrangement and external features of the engine are shown in Figure II-1, with principle physical characteristics listed in Table II-I. The reduction gear shown in Figure II-1 has an overall gear ratio of 10.83:1, providing a propfan speed of 1390 rpm at a constant power turbine output speed of 15,049 rpm. However, parametric weight data is shown in Section III so that other propfan rotational speeds, and gear ratios can be analyzed. An aircraft accessory drive pad is provided on the back of the gearbox to drive an aircraft mounted accessory drive box. Power available at this pad is 500 HP at 8000 rpm. The primary engine mounts are on the gearbox with a hang mount at the rear of the engine. Engine accessories are driven by a bevel drive from the high pressure spool. The control system is integral with the prop-fan and is electro-mechanical. The oil system is integral to the engine and also supplies the prop-fan and reduction gearbox, but is separately filtered and monitored to isolate fault detection in each of these major modules. Engine torque is measured hydraulically from the gear thrust of the power train idler gears in the reduction gearbox.

The gearbox is shown offset, based upon DDA's experience with large turboprop engines. It is offset-up to be consistent with current studies showing a preference to under-the-wing engine mounting. It can also be supplied in the offset-down position.

Performance ratings, sea level static, are listed in Table II-II.



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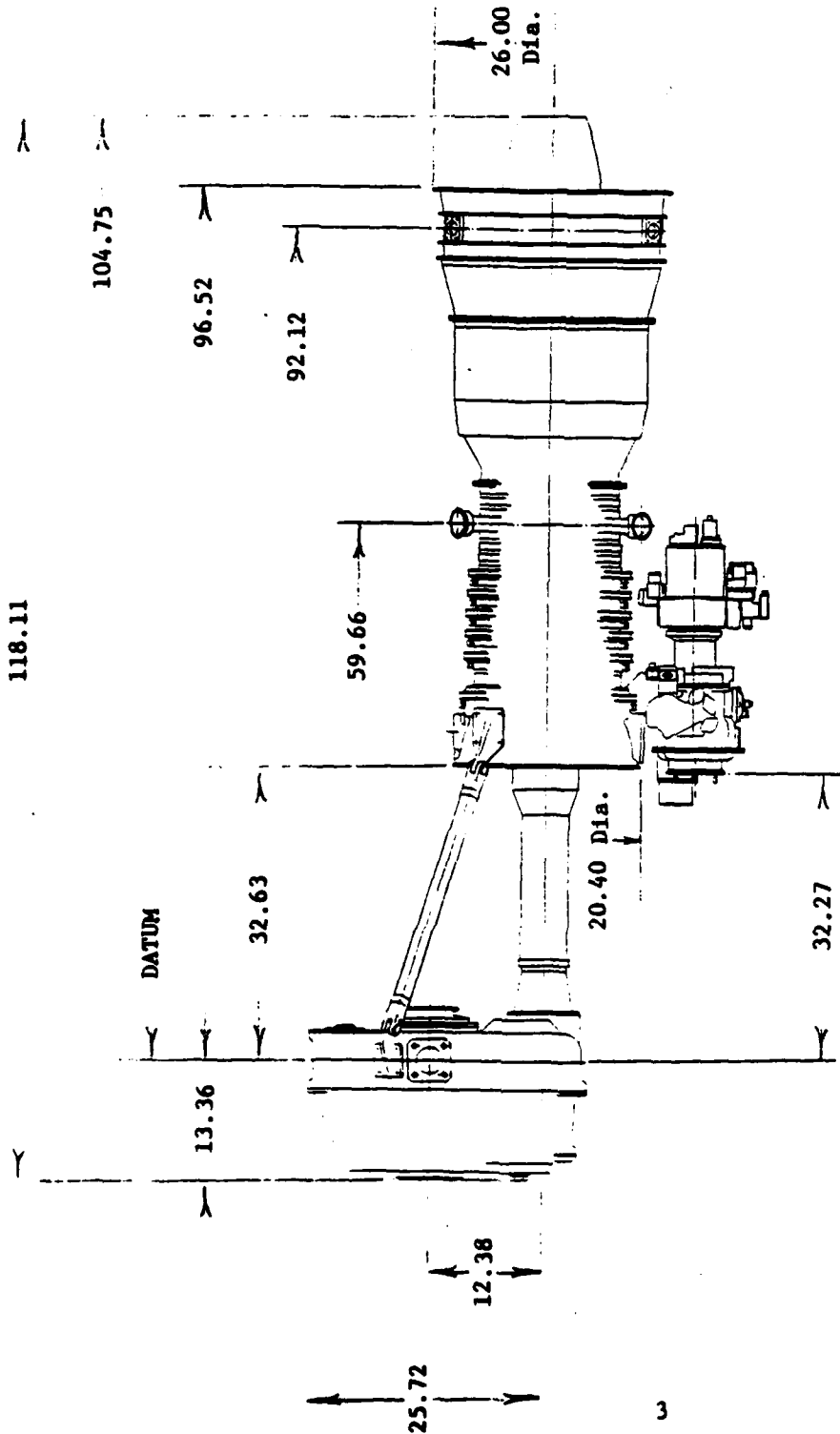


Figure II-1. PD370-37 General Arrangement.



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TABLE II-I

PD 370-37 PHYSICAL CHARACTERISTICS

(Includes Gearbox)

Length (in)	118.11
Max Engine Diameter (in)	26.00
Max Gearbox Offset, Upward (in)	25.72
Dry Weight, lbs.	
Engine	1105
Gearbox, including Interconnecting Struts & Shaft	501
Total	1606

For gear ratios other than 10.83:1 the reduction gearbox dimensions may be scaled as follows:

$$\text{Dim}_{\text{GR}} = \text{Base dim} \times \left(\frac{\text{GR}}{10.83} \right)^{0.33}$$

Sea level performance ratings are summarized in Table II-II.

TABLE II-II

PD 370-37 PERFORMANCE SUMMARY

Sea Level, 0 Kts.

	<u>Standard Day</u>			<u>Hot Day, 89.8°F</u>		
	SHP	SFC	F _N	SHP	SFC	F _N
Take-Off	8450	0.443	1213	7766	0.448	1100
Max. Continuous	7455	0.443	1079	6408	0.461	913



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III. WEIGHTS

The weight of the basic engine, gearbox, interconnecting struts and shaft are given in Table III-I. The gearbox weight is based upon a gear ratio of 10.83:1 which provides a propfan speed of 1390 rpm.

TABLE III-I

PD 370-37 Weights

	<u>Dry</u>	<u>Wet</u> [*]	<u>Installed</u>
Basic Engine, lbs.	1105	1128	1128
Gearbox, lbs.	471	508	508
Interconnecting Struts and Shaft	<u>30</u>	<u>30</u>	<u>30</u>
Total, lbs.	1606	1666	1666

* Includes total amount of oil required for engine and gearbox operation.

For gear ratios other than 10.83:1, the gearbox, interconnecting strut and shaft dry weights may be estimated as follows:

$$\text{Dry Gearbox weight} = 471 \left(\frac{\text{GR}}{10.83} \right)^{0.4}$$

Interconnecting strut and shaft weight = 6.4% of dry gearbox weight.



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IV. STEADY STATE PERFORMANCE

Steady state performance data is tabulated in this section for all points shown in Figure IV-1. Basic engine data is shown for the following assumptions:

- o Uninstalled engine
- o ICAO standard atmosphere except for takeoff which in addition includes an ambient temperature of 89.8°F at standard atmosphere
- o 100% inlet recovery
- o Zero accessory horsepower extraction
- o Zero customer bleed extraction
- o Zero losses due to reduction gear
- o Fuel heating value - 18,400 Btu/lb
- o Estimated average engine performance - No SHP or fuel flow guarantee factors

Sensitivity data is provided for each point so that bleed and duct losses may be estimated as required.

Nomenclature

Nomenclature used in the tabulation of performance is as follows:

MACH	Mach number
SHP	Shaft horsepower
SFC	Specific fuel consumption, lbs/hr/hp
WF	Engine fuel flow, lbs/hr
FN	Net jet thrust, lbs (jet gross thrust - ram drag)
ESHP	Equivalent shaft horsepower (energy in jet stream converted ideally to horsepower and added to SHP)
WCIN	Total inlet corrected airflow, $W\sqrt{\theta_1}/\delta_1$
	where: θ_1 = Engine inlet total temp, °R
	$\frac{518.688}{\theta_1}$



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$$\delta_1 = \frac{\text{Engine inlet total pressure, psi}}{14.696}$$

TNOZ	Jet nozzle total temperature, °R
PNOZ	Jet nozzle total pressure, psi
RC	Compressor pressure ratio
BOT	Burner outlet temperature, °R
NO	Point number

Sensitivity Data

Bleed:

$$\text{SHP, with bleed} = \text{SHP, no bleed} - (\text{DEL SHP})(\% \text{ bleed})$$

$$\text{WF, with bleed} = \text{WF, no bleed} - (\text{DEL WF})(\% \text{ bleed})$$

$$\text{FN, with bleed} = \text{FN, no bleed} - (\text{DEL FN})(\% \text{ bleed})$$

Inlet Recovery:

$$\eta_R = \text{Total pressure actual/Total pressure ideal}$$

$$\text{SHP, with recovery} = \text{SHP, ideal recovery} - (\text{DEL SHP})(1 - \eta_R)(100)$$

$$\text{WF, with recovery} = \text{WF, ideal recovery} (\eta_R)$$

$$\text{FN, with recovery} = \text{FN, ideal recovery} - (\text{DEL FN})(1 - \eta_R)(100)$$

Jet Nozzle Duct Loss:

To estimate thrust loss due to additional duct loss prior to the jet nozzle, use the following equation:

$$\text{FN, with loss} = \text{FN, without loss} - \text{FN, without loss} (K) \left(\frac{\Delta P}{P} \right)$$

where,

o K is obtained for each point from sensitivity data

$$o \frac{\Delta P}{P} = \frac{\text{PTOT, no loss} - \text{PTOT, total loss}}{\text{PTOT, no loss}}$$



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Reduction Gear Loss:

Reduction gear is 99 percent efficient.

Accessory Drive Losses:

Accessory drive power extraction is directly from the accessory drive pad on the reduction gearbox. Reduce SHP to prop-fan by amount of accessory power extraction at each point.

Nozzle Throat Area

The effective nozzle throat area is constant for all conditions at 300.0 in².



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Standard and Hot Day; Takeoff and Maximum Continuous

Altitude ($\text{Ft} \times 10^{-3}$)	MACH Number			
	0	.1	.2	.3
0	X	X	X	X

Standard Day; Maximum Climb, Maximum Continuous and Part Power to Idle

Altitude ($\text{Ft} \times 10^{-3}$)	MACH Number							
	.2	.3	.4	.5	.6	.7	.75	.8
0	X	X	X	X	X	X	X	X
5	X	X	X	X	X			
10	X	X	X	X	X	X	X	X
15		X	X	X	X	X	X	X
20			X	X	X	X	X	X
25			X	X	X	X	X	X
30				X	X	X	X	X
35				X	X	X	X	X
40				X	X	X	X	X
45				X	X	X	X	X

Figure IV-1. Matrix of flight conditions for performance data

DETROIT DIESEL ALLISCA DIVISION
 ZERO BLEED
 EDR 9774
 P0370-37 TURBOPROP
 100 PERCENT RECOVERY
 STD DAY
 C FEET ALTITUDE
 POWER MACH SHP SFC NF FA ESHP WCIN TNDZ PNDZ RC NO
 TO 1 0.0 8550. 0.443 3744. 1213.2 9404. 42.9 1638. 16.82 12.7 0001
 M.C. 0.0 7511. 0.443 3744. 1070.8 8233. 42.9 1547. 16.58 12.7 0002
 TO 2 0.1 7501. 0.443 3744. 947.5 8233. 42.9 1546. 16.59 12.7 0003
 M.C. 0.2 7501. 0.443 3744. 847.5 8233. 42.9 1546. 16.59 12.7 0004
 TO 3 0.3 7501. 0.443 3744. 747.5 8233. 42.9 1546. 16.59 12.7 0005
 M.C. 0.4 7501. 0.443 3744. 647.5 8233. 42.9 1546. 16.59 12.7 0006
 TO 5 0.5 7501. 0.443 3744. 547.5 8233. 42.9 1546. 16.59 12.7 0007
 M.C. 0.6 7501. 0.443 3744. 447.5 8233. 42.9 1546. 16.59 12.7 0008

T AMBIENT = 89.8°F

0 FEET ALTITUDE

TO 1 0.0 7766. 0.443 3744. 1399.5 8595. 43.1 1573. 16.91 12.7 0009
 M.C. 0.0 7766. 0.443 3744. 1399.5 8595. 43.1 1573. 16.91 12.7 0010
 TO 2 0.1 7766. 0.443 3744. 1399.5 8595. 43.1 1573. 16.91 12.7 0011
 M.C. 0.2 7766. 0.443 3744. 1399.5 8595. 43.1 1573. 16.91 12.7 0012
 TO 3 0.3 7766. 0.443 3744. 1399.5 8595. 43.1 1573. 16.91 12.7 0013
 M.C. 0.4 7766. 0.443 3744. 1399.5 8595. 43.1 1573. 16.91 12.7 0014
 TO 5 0.5 7766. 0.443 3744. 1399.5 8595. 43.1 1573. 16.91 12.7 0015
 M.C. 0.6 7766. 0.443 3744. 1399.5 8595. 43.1 1573. 16.91 12.7 0016

DETROIT DIESEL ALLISON DIVISION EDR 9774 PC370-37 TURBOPROP GENERAL MOTORS CORPORATION

SENSITIVITY DATA FOR BLEED, INLET RECOVERY, AND EXHAUST DUCT LOSS

		0 FEET ALTITUDE		0 FEET ALTITUDE					
		PER SHIP	PER DEL	PER DEL	PER DEL	DEL FR	K	NO	
POWER	MACH	DEL SHIP	DEL DEL	DEL DEL	DEL DEL	DEL FR	K	NO	
TO	0.0	14.1	31.1	31.1	31.1	31.1	3.68	0001	
M.C.	0.0	14.1	31.1	31.1	31.1	31.1	3.68	0002	
TO	0.0	14.1	31.1	31.1	31.1	31.1	3.68	0003	
M.C.	0.0	14.1	31.1	31.1	31.1	31.1	3.68	0004	
TO	0.0	14.1	31.1	31.1	31.1	31.1	3.68	0005	
M.C.	0.0	14.1	31.1	31.1	31.1	31.1	3.68	0006	
TO	0.0	14.1	31.1	31.1	31.1	31.1	3.68	0007	
M.C.	0.0	14.1	31.1	31.1	31.1	31.1	3.68	0008	

T AMBIENT = 89.6° F

0 FEET ALTITUDE

		0 FEET ALTITUDE		0 FEET ALTITUDE					
		PER SHIP	PER DEL	PER DEL	PER DEL	DEL FR	K	NO	
POWER	MACH	DEL SHIP	DEL DEL	DEL DEL	DEL DEL	DEL FR	K	NO	
TO	0.0	14.1	31.1	31.1	31.1	31.1	3.68	0009	
M.C.	0.0	14.1	31.1	31.1	31.1	31.1	3.68	0010	
TO	0.0	14.1	31.1	31.1	31.1	31.1	3.68	0011	
M.C.	0.0	14.1	31.1	31.1	31.1	31.1	3.68	0012	
TO	0.0	14.1	31.1	31.1	31.1	31.1	3.68	0013	
M.C.	0.0	14.1	31.1	31.1	31.1	31.1	3.68	0014	
TO	0.0	14.1	31.1	31.1	31.1	31.1	3.68	0015	
M.C.	0.0	14.1	31.1	31.1	31.1	31.1	3.68	0016	

ZERO POWER EXTRACTION

EDR 9774

PD370-37 TURBOPROP

100 PERCENT RECOVERY

STD DAY

0 FEET ALTITUDE

DETROIT DIESEL ALLISON DIVISION

ZERO BLEED

NO

RC

PNOZ

701

MC IN

ESHP

23

34

SFC

DHS

ACH

POWER

1 2 3 4 5 6 7 8 9 10 11 12 13 14 15 16 17 18 19 20 21 22 23 24 25 26 27 28 29 30 31 32 33 34 35 36 37 38 39 40 41 42 43 44 45 46 47 48 49 50 51 52 53 54 55 56 57 58 59 60 61 62 63 64 65 66 67 68 69 70 71 72 73 74 75 76 77 78 79 80 81 82 83 84 85 86 87 88 89 90 91 92 93 94 95 96 97 98 99 100 101 102 103 104 105 106 107 108 109 110 111 112 113 114 115 116 117 118 119 120 121 122 123 124 125 126 127 128 129 130 131 132 133 134 135 136 137 138 139 140 141 142 143 144 145 146 147 148 149 150 151 152 153 154 155 156 157 158 159 160 161 162 163 164 165 166 167 168 169 170 171 172 173 174 175 176 177 178 179 180 181 182 183 184 185 186 187 188 189 190 191 192 193 194 195 196 197 198 199 200 201 202 203 204 205 206 207 208 209 210 211 212 213 214 215 216 217 218 219 220 221 222 223 224 225 226 227 228 229 230 231 232 233 234 235 236 237 238 239 240 241 242 243 244 245 246 247 248 249 250 251 252 253 254 255 256 257 258 259 260 261 262 263 264 265 266 267 268 269 270 271 272 273 274 275 276 277 278 279 280 281 282 283 284 285 286 287 288 289 290 291 292 293 294 295 296 297 298 299 300 301 302 303 304 305 306 307 308 309 310 311 312 313 314 315 316 317 318 319 320 321 322 323 324 325 326 327 328 329 330 331 332 333 334 335 336 337 338 339 340 341 342 343 344 345 346 347 348 349 350 351 352 353 354 355 356 357 358 359 360 361 362 363 364 365 366 367 368 369 370 371 372 373 374 375 376 377 378 379 380 381 382 383 384 385 386 387 388 389 390 391 392 393 394 395 396 397 398 399 400 401 402 403 404 405 406 407 408 409 410 411 412 413 414 415 416 417 418 419 420 421 422 423 424 425 426 427 428 429 430 431 432 433 434 435 436 437 438 439 440 441 442 443 444 445 446 447 448 449 450 451 452 453 454 455 456 457 458 459 460 461 462 463 464 465 466 467 468 469 470 471 472 473 474 475 476 477 478 479 480 481 482 483 484 485 486 487 488 489 490 491 492 493 494 495 496 497 498 499 500 501 502 503 504 505 506 507 508 509 510 511 512 513 514 515 516 517 518 519 520 521 522 523 524 525 526 527 528 529 530 531 532 533 534 535 536 537 538 539 540 541 542 543 544 545 546 547 548 549 550 551 552 553 554 555 556 557 558 559 560 561 562 563 564 565 566 567 568 569 570 571 572 573 574 575 576 577 578 579 580 581 582 583 584 585 586 587 588 589 590 591 592 593 594 595 596 597 598 599 600 601 602 603 604 605 606 607 608 609 610 611 612 613 614 615 616 617 618 619 620 621 622 623 624 625 626 627 628 629 630 631 632 633 634 635 636 637 638 639 640 641 642 643 644 645 646 647 648 649 650 651 652 653 654 655 656 657 658 659 660 661 662 663 664 665 666 667 668 669 670 671 672 673 674 675 676 677 678 679 680 681 682 683 684 685 686 687 688 689 690 691 692 693 694 695 696 697 698 699 700 701 702 703 704 705 706 707 708 709 710 711 712 713 714 715 716 717 718 719 720 721 722 723 724 725 726 727 728 729 730 731 732 733 734 735 736 737 738 739 740 741 742 743 744 745 746 747 748 749 750 751 752 753 754 755 756 757 758 759 760 761 762 763 764 765 766 767 768 769 770 771 772 773 774 775 776 777 778 779 780 781 782 783 784 785 786 787 788 789 790 791 792 793 794 795 796 797 798 799 800 801 802 803 804 805 806 807 808 809 810 811 812 813 814 815 816 817 818 819 820 821 822 823 824 825 826 827 828 829 830 831 832 833 834 835 836 837 838 839 840 841 842 843 844 845 846 847 848 849 850 851 852 853 854 855 856 857 858 859 860 861 862 863 864 865 866 867 868 869 870 871 872 873 874 875 876 877 878 879 880 881 882 883 884 885 886 887 888 889 890 891 892 893 894 895 896 897 898 899 900 901 902 903 904 905 906 907 908 909 910 911 912 913 914 915 916 917 918 919 920 921 922 923 924 925 926 927 928 929 930 931 932 933 934 935 936 937 938 939 940 941 942 943 944 945 946 947 948 949 950 951 952 953 954 955 956 957 958 959 960 961 962 963 964 965 966 967 968 969 970 971 972 973 974 975 976 977 978 979 980 981 982 983 984 985 986 987 988 989 990 991 992 993 994 995 996 997 998 999 1000 1001 1002 1003 1004 1005 1006 1007 1008 1009 1010 1011 1012 1013 1014 1015 1016 1017 1018 1019 1020 1021 1022 1023 1024 1025 1026 1027 1028 1029 1030 1031 1032 1033 1034 1035 1036 1037 1038 1039 104

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WM173

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DETROIT DIESEL ALLISON DIVISION

100 PERCENT RECOVERY
SIX DAY

ZERO BLEED

[illegible]

GENERAL MOTORS CORPORATION
ZERO POWER EXTRACTION

DETROIT DIESEL ALLISON DIVISION

P0370-37 TURBOPROP

ZERO BLEED

100 PERCENT RECOVERY

STD DAY

15000 FEET ALTITUDE

[illegible]

GENERAL MOTORS CORPORATION

ZERO POWER EXTRACTION

30000 FEET ALTITUDE

20

18

EDR 9774

PO370-37 TUNABPROP

100 PERCENT RECOVERY

STD DAY

35000 FEET ALTITUDE

GENERAL MOTORS CORPORATION

ZERO POWER EXTRACTION

[illegible]

DETROIT DIESEL ALLISON DIVISION

EDR 9774

GENERAL MOTORS CORPORATION

PD370-37 TURBOPROP

ZERO BLEED

100 PERCENT RECOVERY

ZERO POWER EXTRACTION

STD DAY

40000 FEET ALTITUDE

POWER	MACH	SHP	SFC	WF	FN	ESHP	WCIN	TNOZ	PNOZ	RC	NO
CLIMB M.C.	00.50	568	52900	309	00594	007	44.6	11.1	375	098	0377
	00.50	509	40000	989	11671	179	44.6	16.0	333	433	0379
	00.50	471	39000	716	11111	111	44.6	22.0	333	433	0380
	00.50	409	32000	508	11801	06	44.6	25.0	333	433	0381
	00.50	358	22000	358	11811	00	44.6	27.0	333	433	0382
	00.60	338	40000	562	11811	00	44.6	31.0	333	433	0383
	00.60	328	39000	535	11811	00	44.6	31.0	333	433	0384
	00.60	328	38000	535	11811	00	44.6	31.0	333	433	0385
	00.60	328	37000	535	11811	00	44.6	31.0	333	433	0386
	00.60	328	36000	535	11811	00	44.6	31.0	333	433	0387
	00.60	328	35000	535	11811	00	44.6	31.0	333	433	0388
	00.60	328	34000	535	11811	00	44.6	31.0	333	433	0389
	00.60	328	33000	535	11811	00	44.6	31.0	333	433	0390
	00.60	328	32000	535	11811	00	44.6	31.0	333	433	0391
	00.60	328	31000	535	11811	00	44.6	31.0	333	433	0392
	00.60	328	30000	535	11811	00	44.6	31.0	333	433	0393
	00.60	328	29000	535	11811	00	44.6	31.0	333	433	0394
	00.60	328	28000	535	11811	00	44.6	31.0	333	433	0395
	00.60	328	27000	535	11811	00	44.6	31.0	333	433	0396
	00.60	328	26000	535	11811	00	44.6	31.0	333	433	0397
	00.60	328	25000	535	11811	00	44.6	31.0	333	433	0398
	00.60	328	24000	535	11811	00	44.6	31.0	333	433	0399
	00.60	328	23000	535	11811	00	44.6	31.0	333	433	0400
	00.60	328	22000	535	11811	00	44.6	31.0	333	433	0401
	00.60	328	21000	535	11811	00	44.6	31.0	333	433	0402
	00.60	328	20000	535	11811	00	44.6	31.0	333	433	0403
	00.60	328	19000	535	11811	00	44.6	31.0	333	433	0404
	00.60	328	18000	535	11811	00	44.6	31.0	333	433	0405
	00.60	328	17000	535	11811	00	44.6	31.0	333	433	0406
	00.60	328	16000	535	11811	00	44.6	31.0	333	433	0407
	00.60	328	15000	535	11811	00	44.6	31.0	333	433	0408
	00.60	328	14000	535	11811	00	44.6	31.0	333	433	0409
	00.60	328	13000	535	11811	00	44.6	31.0	333	433	0410
	00.60	328	12000	535	11811	00	44.6	31.0	333	433	0411
	00.60	328	11000	535	11811	00	44.6	31.0	333	433	0412
	00.60	328	10000	535	11811	00	44.6	31.0	333	433	0413
	00.60	328	9000	535	11811	00	44.6	31.0	333	433	0414
	00.60	328	8000	535	11811	00	44.6	31.0	333	433	0415
	00.60	328	7000	535	11811	00	44.6	31.0	333	433	0416

DETROIT DIESEL ALLISON DIVISION

EDR 9774

GENERAL MOTORS CORPORATION

ZERO BLEED

P0370-37 TURBOPROP

ZERO POWER EXTRACTION

100 PERCENT RECOVERY

STD DAY

45000 FEET ALTITUDE

POWER	MACH	SHF	SFC	WF	FN	ESHP	WCIN	TNOZ	PN0Z	RC	NO
CLIMB M.C.	0.50	1774.	412	736.	9.5	2051.	6.6	511.	65	0	17
	0.50	1794.	409	712.	152.	2041.	44.	114.	22	43	19
	0.50	1814.	406	688.	154.	2031.	44.	135.	40	13	20
	0.50	1834.	403	664.	156.	2021.	44.	156.	37	13	21
	0.50	1854.	400	640.	158.	2011.	44.	177.	34	13	22
	0.50	1874.	397	616.	160.	2001.	44.	198.	31	13	23
	0.50	1894.	394	592.	162.	1991.	44.	219.	28	13	24
	0.60	1914.	391	568.	164.	1981.	44.	240.	25	13	25
	0.60	1934.	388	544.	166.	1971.	44.	261.	22	13	26
	0.60	1954.	385	520.	168.	1961.	44.	282.	19	13	27
	0.60	1974.	382	496.	170.	1951.	44.	303.	16	13	28
	0.60	1994.	379	472.	172.	1941.	44.	324.	13	13	29
	0.60	2014.	376	448.	174.	1931.	44.	345.	10	13	30
	0.60	2034.	373	424.	176.	1921.	44.	366.	7	13	31
	0.60	2054.	370	400.	178.	1911.	44.	387.	4	13	32
	0.70	2074.	367	376.	180.	1901.	44.	408.	1	13	33
	0.70	2094.	364	352.	182.	1891.	44.	429.	0	13	34
	0.70	2114.	361	328.	184.	1881.	44.	450.	0	13	35
	0.70	2134.	358	304.	186.	1871.	44.	471.	0	13	36
	0.70	2154.	355	280.	188.	1861.	44.	492.	0	13	37
	0.70	2174.	352	256.	190.	1851.	44.	513.	0	13	38
	0.70	2194.	349	232.	192.	1841.	44.	534.	0	13	39
	0.70	2214.	346	208.	194.	1831.	44.	555.	0	13	40
	0.70	2234.	343	184.	196.	1821.	44.	576.	0	13	41
	0.70	2254.	340	160.	198.	1811.	44.	597.	0	13	42
	0.70	2274.	337	136.	200.	1801.	44.	618.	0	13	43
	0.70	2294.	334	112.	202.	1791.	44.	639.	0	13	44
	0.70	2314.	331	88.	204.	1781.	44.	660.	0	13	45
	0.70	2334.	328	64.	206.	1771.	44.	681.	0	13	46
	0.70	2354.	325	40.	208.	1761.	44.	702.	0	13	47
	0.70	2374.	322	16.	210.	1751.	44.	723.	0	13	48
	0.70	2394.	319	0.	212.	1741.	44.	744.	0	13	49
	0.80	2414.	316	0.	214.	1731.	44.	765.	0	13	50
	0.80	2434.	313	0.	216.	1721.	44.	786.	0	13	51
	0.80	2454.	310	0.	218.	1711.	44.	807.	0	13	52
	0.80	2474.	307	0.	220.	1701.	44.	828.	0	13	53
	0.80	2494.	304	0.	222.	1691.	44.	849.	0	13	54
	0.80	2514.	301	0.	224.	1681.	44.	870.	0	13	55
	0.80	2534.	298	0.	226.	1671.	44.	891.	0	13	56
	0.80	2554.	295	0.	228.	1661.	44.	912.	0	13	57
	0.80	2574.	292	0.	230.	1651.	44.	933.	0	13	58
	0.80	2594.	289	0.	232.	1641.	44.	954.	0	13	59
	0.80	2614.	286	0.	234.	1631.	44.	975.	0	13	60

GENERAL MOTORS CORPORATION

0 FEET ALTITUDE

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EDR 9774

PD370-37 TURBOPROP

GENERAL MOTORS CORPORATION

SENSITIVITY DATA FOR BLEED, INLET RECOVERY, AND EXHAUST DUCT LOSS

5000 FEET ALTITUDE

POWER
CLIMB
M.C.

CLIMB
M.C.

CLIMB
M.C.

CLIMB
M.C.

CLIMB
M.C.

MACH

DEL SHIP INTL DEL BFM

DEL SHIP PER DEL MFL DEL FR

K

NO

DETROIT DIESEL ALLISON DIVISION

SENSITIVITY DATA FOR BLEED, INLET RECOVERY, AND EXHAUST DUCT LOSS

10000 FEET ALTITUDE

[illegible]

EDR 9774

P0370-37 TURBOPROP

GENERAL MOTORS CORPORATION

SENSITIVITY DATA FOR BLEED, INLET RECOVERY, AND EXHAUST DUCT LOSS

15000 FEET ALTITUDE

[illegible]

GENERAL MOTORS CORPORATION

SENSITIVITY DATA FOR BLEED, INLET RECOVERY, AND EXHAUST DUCT LOSS

20000 FEET ALTITUDE

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SENSITIVITY DATA FOR BLEED, INLET RECOVERY, AND EXHAUST DUCT LOSS

25000 FEET ALTITUDE

27

DETROIT DIESEL ALLISON DIVISION

EDR 5774

GENERAL MOTORS CORPORATION

PD370-37 TURBOPROP

SENSITIVITY DATA FOR BLEED, INLET RECOVERY, AND EXHAUST DUCT LOSS

30000 FEET ALTITUDE

POWER	MACH	DEL-SHP	INLET DEL-FN	DEL-SHP	PER DEL-BF	DEL-FN	K	NO
CLIMB M.C.	0.50	7.2	3.0	46.1	1.1	0.8	1.1	0297
	0.50	5.5	2.8	45.1	1.0	0.8	1.0	0298
	0.50	4.2	2.6	44.1	1.0	0.8	1.0	0300
	0.50	3.2	2.4	43.1	1.0	0.8	1.0	0301
	0.50	2.5	2.2	42.1	1.0	0.8	1.0	0302
	0.50	2.0	2.0	41.1	1.0	0.8	1.0	0303
	0.50	1.6	1.8	40.1	1.0	0.8	1.0	0304
	0.50	1.3	1.6	39.1	1.0	0.8	1.0	0305
	0.50	1.1	1.4	38.1	1.0	0.8	1.0	0306
	0.50	0.9	1.2	37.1	1.0	0.8	1.0	0307
	0.50	0.8	1.0	36.1	1.0	0.8	1.0	0308
	0.50	0.7	0.9	35.1	1.0	0.8	1.0	0309
	0.50	0.6	0.8	34.1	1.0	0.8	1.0	0310
	0.50	0.5	0.7	33.1	1.0	0.8	1.0	0311
	0.50	0.4	0.6	32.1	1.0	0.8	1.0	0312
	0.50	0.3	0.5	31.1	1.0	0.8	1.0	0313
	0.50	0.2	0.4	30.1	1.0	0.8	1.0	0314
	0.50	0.1	0.3	29.1	1.0	0.8	1.0	0315
	0.50	0.1	0.2	28.1	1.0	0.8	1.0	0316
	0.50	0.1	0.1	27.1	1.0	0.8	1.0	0317
	0.50	0.1	0.1	26.1	1.0	0.8	1.0	0318
	0.50	0.1	0.1	25.1	1.0	0.8	1.0	0319
	0.50	0.1	0.1	24.1	1.0	0.8	1.0	0320
	0.50	0.1	0.1	23.1	1.0	0.8	1.0	0321
	0.50	0.1	0.1	22.1	1.0	0.8	1.0	0322
	0.50	0.1	0.1	21.1	1.0	0.8	1.0	0323
	0.50	0.1	0.1	20.1	1.0	0.8	1.0	0324
	0.50	0.1	0.1	19.1	1.0	0.8	1.0	0325
	0.50	0.1	0.1	18.1	1.0	0.8	1.0	0326
	0.50	0.1	0.1	17.1	1.0	0.8	1.0	0327
	0.50	0.1	0.1	16.1	1.0	0.8	1.0	0328
	0.50	0.1	0.1	15.1	1.0	0.8	1.0	0329
	0.50	0.1	0.1	14.1	1.0	0.8	1.0	0330
	0.50	0.1	0.1	13.1	1.0	0.8	1.0	0331
	0.50	0.1	0.1	12.1	1.0	0.8	1.0	0332
	0.50	0.1	0.1	11.1	1.0	0.8	1.0	0333
	0.50	0.1	0.1	10.1	1.0	0.8	1.0	0334
	0.50	0.1	0.1	9.1	1.0	0.8	1.0	0335
	0.50	0.1	0.1	8.1	1.0	0.8	1.0	0336
	0.50	0.1	0.1	7.1	1.0	0.8	1.0	0337
	0.50	0.1	0.1	6.1	1.0	0.8	1.0	0338
	0.50	0.1	0.1	5.1	1.0	0.8	1.0	0339
	0.50	0.1	0.1	4.1	1.0	0.8	1.0	0340
	0.50	0.1	0.1	3.1	1.0	0.8	1.0	0341
	0.50	0.1	0.1	2.1	1.0	0.8	1.0	0342
	0.50	0.1	0.1	1.1	1.0	0.8	1.0	0343
	0.50	0.1	0.1	0.1	1.0	0.8	1.0	0344

DETROIT DIESEL ALLISON DIVISION

EDR 9774

GENERAL MOTORS CORPORATION

P0370-37 TURBOPROP

SENSITIVITY DATA FOR BLEED, INLET RECOVERY, AND EXHAUST DUCT LOSS

35000 FEET ALTITUDE

POWER CLIMB M.C.	MACH	PER SHP DEL SHP	INLET DEL IN	PER SHP DEL SHP	BLEED DEL BLEED	K	NO
0.50	0.50	36.3	6.1	37.7	3.1	4.1	0337
0.50	0.50	35.4	5.5	36.0	7.3	4.2	0338
0.50	0.50	34.2	4.3	35.0	7.6	4.3	0339
0.50	0.50	33.0	3.1	33.5	7.4	4.4	0340
0.50	0.50	32.2	1.6	32.9	9.9	4.5	0341
0.50	0.50	31.5	0.5	31.8	9.2	4.6	0342
0.60	0.60	30.9	0.2	30.9	8.0	4.6	0343
0.60	0.60	30.0	0.0	30.0	7.1	4.7	0344
0.60	0.60	29.2	0.0	29.5	6.8	4.8	0345
0.60	0.60	28.4	0.0	28.0	6.0	4.9	0346
0.70	0.70	27.6	0.0	27.1	5.3	5.0	0347
0.70	0.70	26.8	0.0	26.3	4.7	5.1	0348
0.70	0.70	26.0	0.0	25.5	4.1	5.2	0349
0.70	0.70	25.2	0.0	24.7	3.6	5.3	0350
0.70	0.70	24.4	0.0	23.9	3.1	5.4	0351
0.70	0.70	23.6	0.0	23.1	2.6	5.5	0352
0.70	0.70	22.8	0.0	22.3	2.1	5.6	0353
0.70	0.70	22.0	0.0	21.5	1.6	5.7	0354
0.70	0.70	21.2	0.0	20.7	1.1	5.8	0355
0.70	0.70	20.4	0.0	19.9	0.6	5.9	0356
0.70	0.70	19.6	0.0	19.1	0.1	6.0	0357
0.70	0.70	18.8	0.0	18.3	0.0	6.1	0358
0.70	0.70	18.0	0.0	17.5	0.0	6.2	0359
0.70	0.70	17.2	0.0	16.7	0.0	6.3	0360
0.70	0.70	16.4	0.0	15.9	0.0	6.4	0361
0.70	0.70	15.6	0.0	15.1	0.0	6.5	0362
0.70	0.70	14.8	0.0	14.3	0.0	6.6	0363
0.70	0.70	14.0	0.0	13.5	0.0	6.7	0364
0.70	0.70	13.2	0.0	12.7	0.0	6.8	0365
0.70	0.70	12.4	0.0	11.9	0.0	6.9	0366
0.70	0.70	11.6	0.0	11.1	0.0	7.0	0367
0.70	0.70	10.8	0.0	10.3	0.0	7.1	0368
0.70	0.70	10.0	0.0	9.5	0.0	7.2	0369
0.70	0.70	9.2	0.0	8.7	0.0	7.3	0370
0.70	0.70	8.4	0.0	7.9	0.0	7.4	0371
0.70	0.70	7.6	0.0	7.1	0.0	7.5	0372
0.70	0.70	6.8	0.0	6.3	0.0	7.6	0373
0.70	0.70	6.0	0.0	5.5	0.0	7.7	0374
0.70	0.70	5.2	0.0	4.7	0.0	7.8	0375
0.70	0.70	4.4	0.0	3.9	0.0	7.9	0376
0.70	0.70	3.6	0.0	3.1	0.0	8.0	0377
0.70	0.70	2.8	0.0	2.3	0.0	8.1	0378
0.70	0.70	2.0	0.0	1.5	0.0	8.2	0379
0.70	0.70	1.2	0.0	0.7	0.0	8.3	0380
0.70	0.70	0.4	0.0	0.0	0.0	8.4	0381
0.70	0.70	0.0	0.0	0.0	0.0	8.5	0382
0.70	0.70	0.0	0.0	0.0	0.0	8.6	0383
0.70	0.70	0.0	0.0	0.0	0.0	8.7	0384
0.70	0.70	0.0	0.0	0.0	0.0	8.8	0385
0.70	0.70	0.0	0.0	0.0	0.0	8.9	0386
0.70	0.70	0.0	0.0	0.0	0.0	9.0	0387
0.70	0.70	0.0	0.0	0.0	0.0	9.1	0388
0.70	0.70	0.0	0.0	0.0	0.0	9.2	0389
0.70	0.70	0.0	0.0	0.0	0.0	9.3	0390
0.70	0.70	0.0	0.0	0.0	0.0	9.4	0391
0.70	0.70	0.0	0.0	0.0	0.0	9.5	0392
0.70	0.70	0.0	0.0	0.0	0.0	9.6	0393
0.70	0.70	0.0	0.0	0.0	0.0	9.7	0394
0.70	0.70	0.0	0.0	0.0	0.0	9.8	0395
0.70	0.70	0.0	0.0	0.0	0.0	9.9	0396
0.70	0.70	0.0	0.0	0.0	0.0	10.0	0397

EDR 9774

GENERAL MOTORS CORPORATION

FD-370-37 TUNBOPROP

SENSITIVITY DATA FOR BLEED, INLET RECOVERY, AND EXHAUST DUCT LOSS

45000 FEET ALTITUDE

[illegible]